NUMBER: M20-49

DATE: 6-5-80

SUBJECT:

MODIFICATION FROM PRESSURE TRANSDUCER SYSTEM TO TURBINE MEASURING SYSTEM FOR FUEL FLOW, M20K AIRCRAFT

MODELS/ SERIAL NUMBERS AFFECTED:

M20K - S/N 25-0001 - 25-0171

TIME OF COMPLIANCE:

At next scheduled maintenance or scheduled as kits are available from Mooney Aircraft Company.

INTRODUCTION:

M20K aircraft, S/N's shown above, were manufactured with a pressure transducer fuel flow gauge readout system. Current production aircraft are using a turbine transducer measuring fuel flow system and is considered to be a better method of determining actual fuel flow required by the engine.

INSTRUCTIONS:

- The illustrations and figures contained in Kit No. 940005-501 depict locations and part numbers of all components required to modify the system to current production configuration. The pressure transducer system will have to be removed completely and the new wires, tubes, brackets, transducers, etc. installed and secured per the figures attached.
- When the turbine transducer and necessary components are installed a leak check should be performed prior to starting the engine.
- If engine has been set-up per Teledyne Continental Motors (T.C.M.) specification, S.B. M79-8 procedure, proceed with calibration of new indicator per Mooney Service Instruction M20-48, dated 2-20-80.
- 4. If the engine has not been set up per above procedures within the past 100 hours accomplish this setup prior to calibration of fuel flow gage. Use 5.82 pounds per gallon as conversion factor.
- 5. Enter statement of compliance in Engine and Aircraft Log Books prior to returning aircraft to service.

S. I. M20-49 Page 2 of 3

REFERENCE DATA:

 Service Instruction M20-48 dated 2-20-80, Calibration of Fuel Flow Gage Used W/Turbine System.

Mooney Aircraft Drawing No. 940005.

PARTS LIST:

See attached sheets, kit P/N 940005-501, Parts List.

FIGURES/

TABLES:

See Mooney Drawing 940005-000 for complete instructions for installation of this modification.

Drawing No.: 940005 Revision Letter: Page No.:							
		No,Reg.FerNote		PART NO.	PART NAME	PAGEIT	
				940005	Drawing		
<u> </u>		1		940005-1	Doubler		
				AN929-4J	Cap, Fig. No. 4		
1.		1.		AN929-3J	Cap, Fig. No. 3		
<u> </u>		1		660233-13	Brkt		
<u> </u>		2		AN3-12A	Bolt		
		1		MS51500B4-4 AN816-4-4	Alternate Mipple		
				6 <u>60</u> 233-509	<u>Fubc</u> Assy		
<u> </u>		1		660233-511			
				MS51508B-4-4	Alternate Fibov	1	
		4		AN363-1032	Nu+	1	
				AN515-6BR-12	· · · · · · · · · · · · · · · · · · ·	1	
		2		AN960-10L	Washer	1	
		2		AN960-10	Washer	1.	
		1		AN526-10R6	Screw	1	
		1		(32622-6) 880030-501	(Aerosonic Corp) Transducer	1	
		1		P]F-30-4	Tubing, Insulating	1	
		1		C10Y-34-6S	Clamp	1	
		1		Cl0x-34-sF	Clamp	1	
						2	
		4			Avex Rivets	2	
		10			Caps, Wire (AMP)	2.	
		6		32446	Disconnects, Knife Bld,	2.	
				651035-0107	Indicator	2.	
		1	 		Harness, Wire	2	
		20	- 		Fires <u>leeve</u> Fy-Rap (TY-3)	2	
		1		2650-28	Safety Wire "Stratoflex"	2	
	*	X	- 	AN995C20		28	
<u> </u>		x		AN995C32	Safety Wire	29	

MOONEY AIRCRAFT CORPORATION

Engry Form 1120

ENGINE SET-UI

	Adjustments							
Data	DATE	DATE	DATE	DATE	DATE			
IDLE PERFORMANCE								
0/M Fuel Press. (6.25 + .25 PSI) (At 700 + 25 RPM)								
Idle Mix. Rise (75 - 100 RPM) (At 700 ± 25 RPM)								
Idla Speed (700 ± 25 RPM)								
FULL POWER PERFORMANCE								
RPM (2700 + 0/-25)								
P/P SHIP P/P TEST (143-145 #/Hr)			!		<u></u>			
GROUND CRITICAL			<u></u>					
PRESSURE ALT. (29.92*tg)								
KAK								
PLIGHT CRITICAL								
PRESS. ALT. (29.92 Hg)		 	 					
*/ (SBIP/TEST) (135-1454/hr)			<u> </u>					
DEMSITY ALT.				[

FLIGHT CRITICAL DATA SHEET

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PRESS ALT	OAT	TIT	F/F F/F	H P H P	я р м я р н	CAL. INST. SHIP	REMARKS
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* ADDITIONAL INFORMATION: